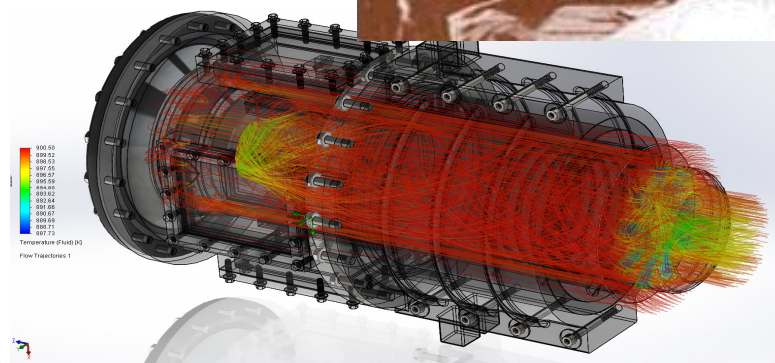
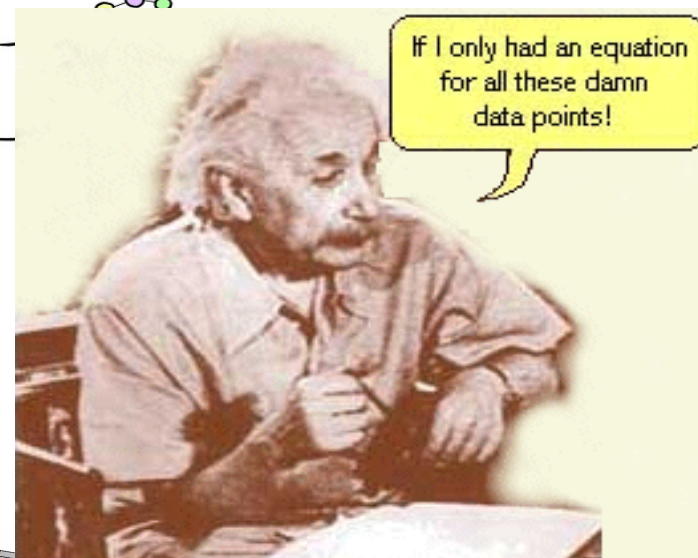
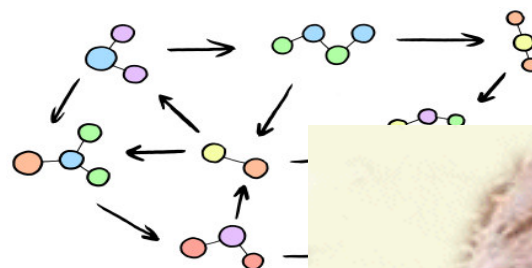


*Scuola Politecnica e delle Scienze di Base
Università Federico II, Napoli
12 ottobre 2019*

Le nuove possibilità della combustione a gravità zero

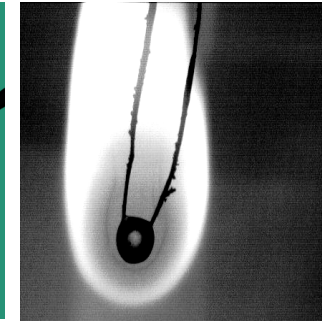
Patrizio Massoli
Istituto Motori – CNR , Napoli, Italy

- The critical point is in the fact that the chemical models have to be used in connection with very heavy CFD simulations
- Also for single component fuels, models with thousands reactions are used to describe the evolution of matter. This approach becomes impracticable when practical fuels with hundreds of compounds are investigated
- Surrogates are considered also for the evaporation phase to simplify the overall approach
- Lumped or reduced chemical schemes based on surrogate fuels are utilized to render acceptable the total computational cost.



Rationale: CFD and Chemical Kinetics

- Simplified fluid-dynamics, heat and mass transfer, to study the fundamental processes during heating and combustion of fuel droplets
- Simplified fuels, the surrogates, that well represent practical fuels in terms of composition, thermo-physical or combustion properties
 - Decane: jet A fuel surrogate;
 - 1-hexanol: renewable long chain alcohol for conventional fuel blends or co-solvent for biodiesel mixtures;
 - heptane and ethanol: light sooting and non sooting fuels



normal g



μg

Rationale: Why μg?

**Experimentation in simplified boundary condition on simplified fuels:
the key to link fundamental processes to real world**



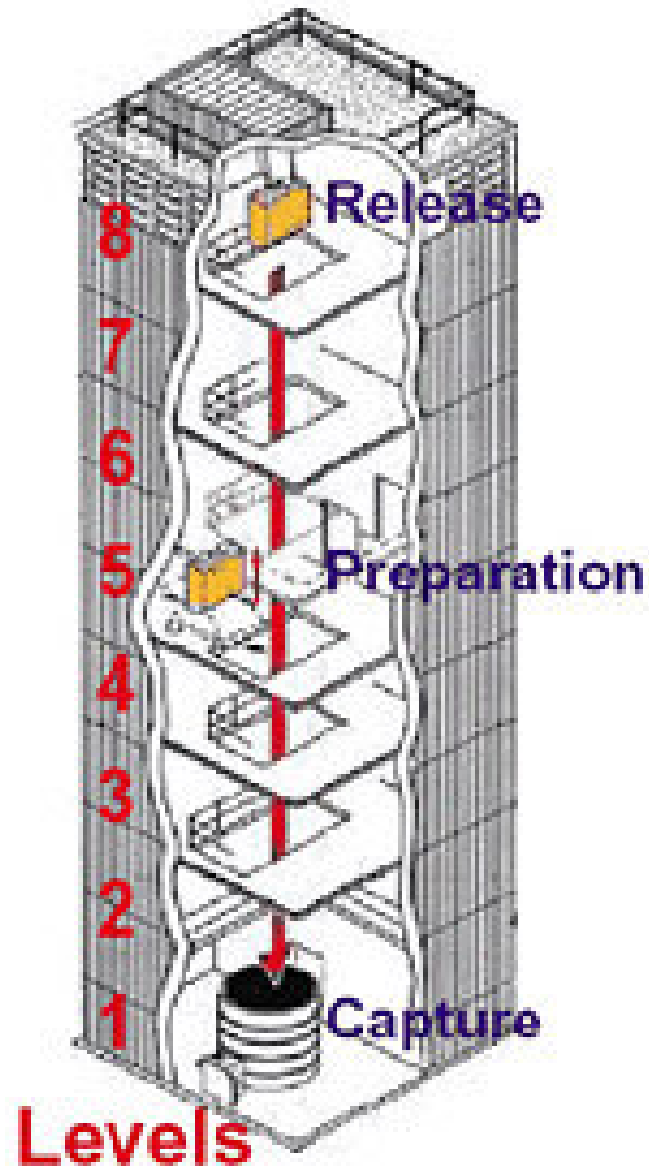
$$m \times 9.8 \text{ ms}^{-2}$$



we can remove the effects of gravity on an object either by placing it in freefall or by recreating freefall conditions

How to obtain 0-gravity

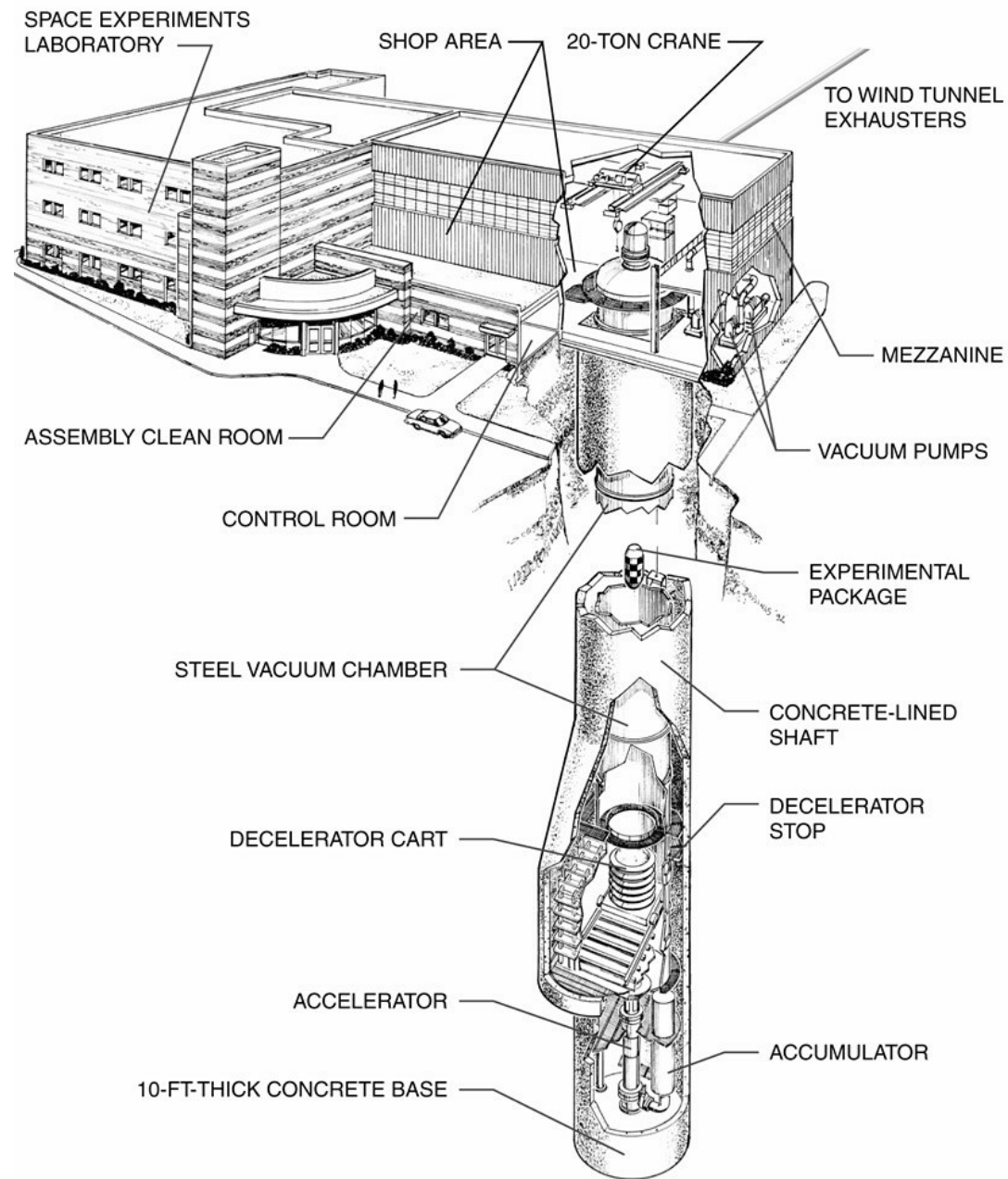




GRC NASA 2.2 seconds drop tower

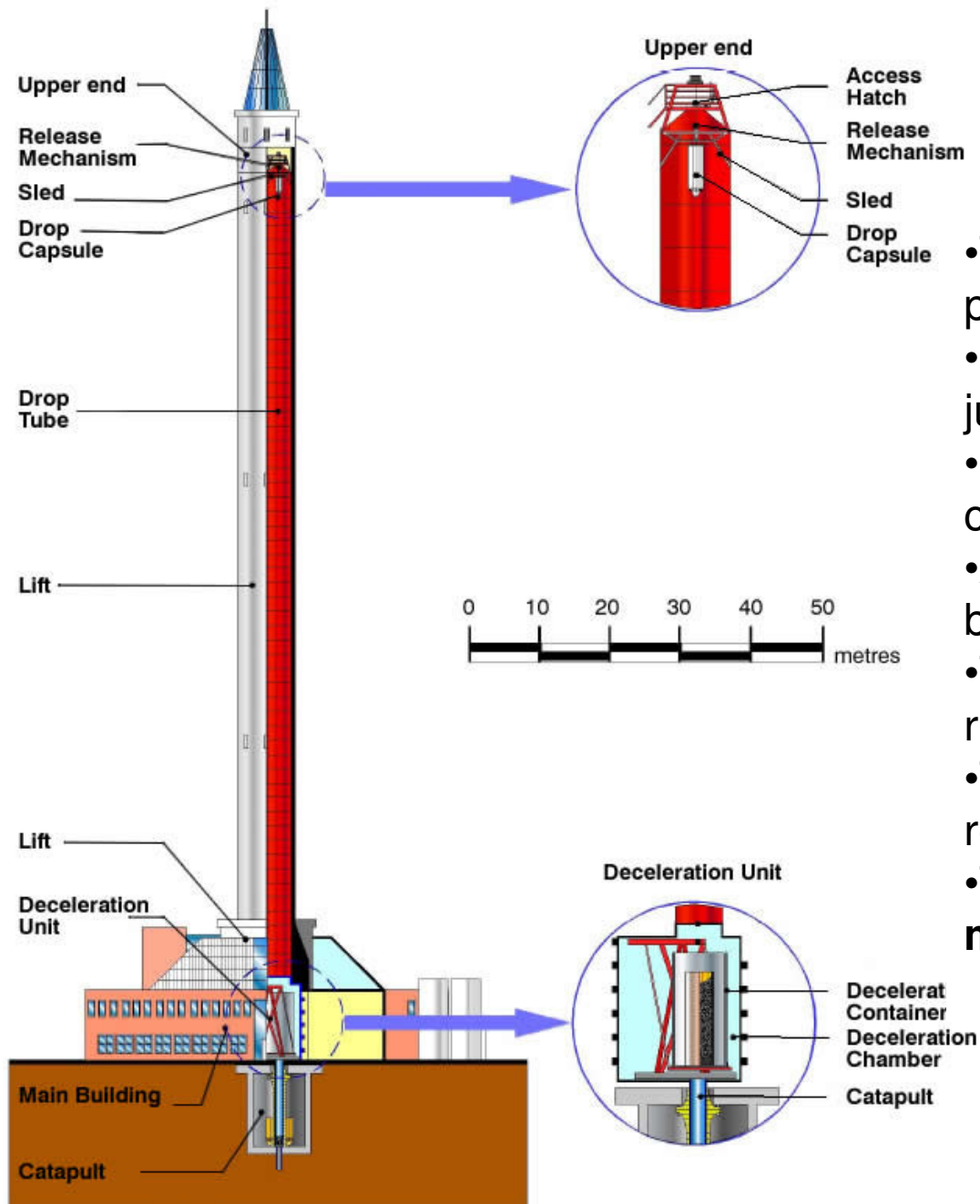
- Microgravity Duration: 2.2 seconds
- Free Fall Distance: 79 feet (24 m)
- Gravitational Acceleration: $< 0.001\text{ g}$
- Mean Deceleration: 15 g
- Peak Deceleration: 30 g

Drop Tower

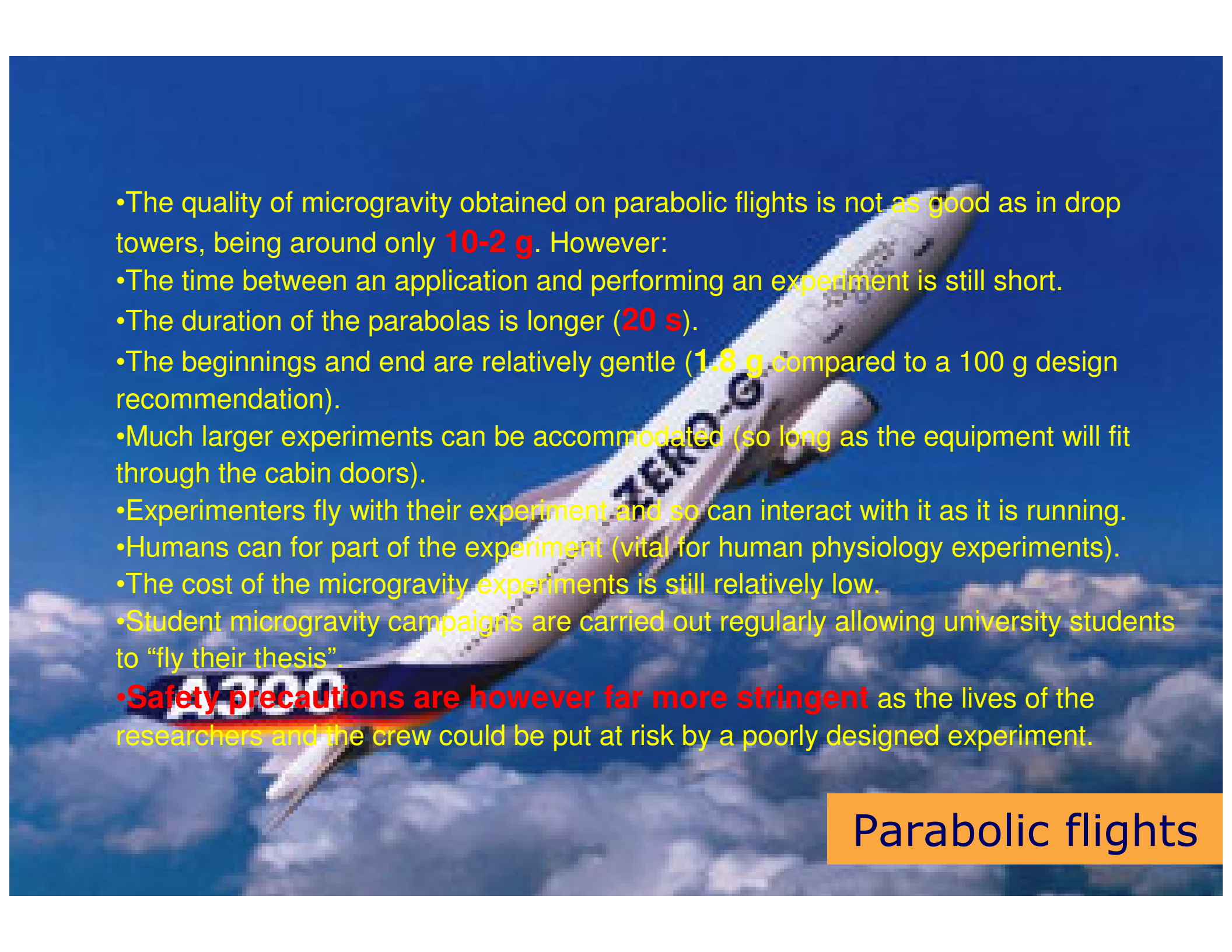


Zero Gravity Research Facility (Zero-G)	
Operational Parameters	
Microgravity Duration:	5.18 seconds
Free Fall Distance:	432 feet (132 m)
Gravitational Acceleration:	<0.00001 g
Mean Deceleration:	35 g
Peak Deceleration:	65 g
Vacuum Level:	0.05 torr

CD-92 - 60458
 BOTTOM OF SHAFT 510 FT BELOW GROUND LEVEL



- The time between an application and performing an experiment is short.
- Excellent standard of microgravity, just **10⁻⁵ g** is considered standard.
- **4.74 seconds or 9.48 seconds** (with catapult: 30 g) of microgravity
- Access to the experiment shortly before and after the drop.
- Turnaround time is quick so several runs can be made in just a few days.
- There are minimal safety requirements.
- The **cost is low compared to other microgravity platforms.**

- 
- The quality of microgravity obtained on parabolic flights is not as good as in drop towers, being around only **10⁻² g**. However:
 - The time between an application and performing an experiment is still short.
 - The duration of the parabolas is longer (**20 s**).
 - The beginnings and end are relatively gentle (**1.8 g** compared to a 100 g design recommendation).
 - Much larger experiments can be accommodated (so long as the equipment will fit through the cabin doors).
 - Experimenters fly with their experiment and so can interact with it as it is running.
 - Humans can for part of the experiment (vital for human physiology experiments).
 - The cost of the microgravity experiments is still relatively low.
 - Student microgravity campaigns are carried out regularly allowing university students to “fly their thesis”.
 - Safety precautions are however far more stringent** as the lives of the researchers and the crew could be put at risk by a poorly designed experiment.

Parabolic flights



ISS Orbit: 450 km

Space Shuttle Orbit: 300 km

MAXUS
705 km

Experimentus
260 km

MiniTEXUS
140 km

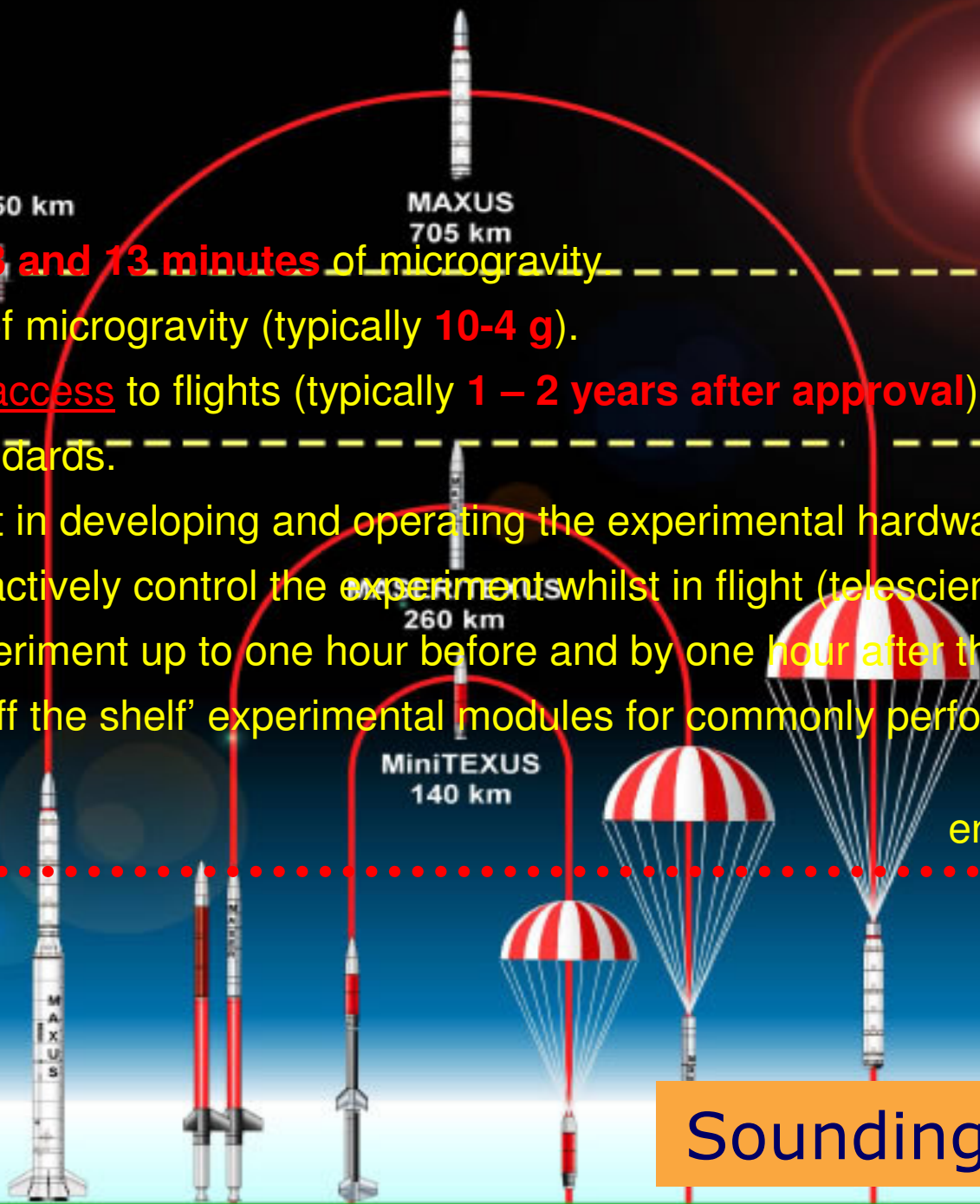
- Provide between **3 and 13 minutes** of microgravity
- A good standard of microgravity (typically **10⁻⁴ g**).
- A relatively quick access to flights (typically **1 – 2 years after approval**).
- Simple safety standards.
- Direct involvement in developing and operating the experimental hardware.
- The possibility to actively control the experiment whilst in flight (telescience).
- Access to the experiment up to one hour before and by one hour after the flight.
- A wide range of 'off the shelf' experimental modules for commonly performed experiment types.

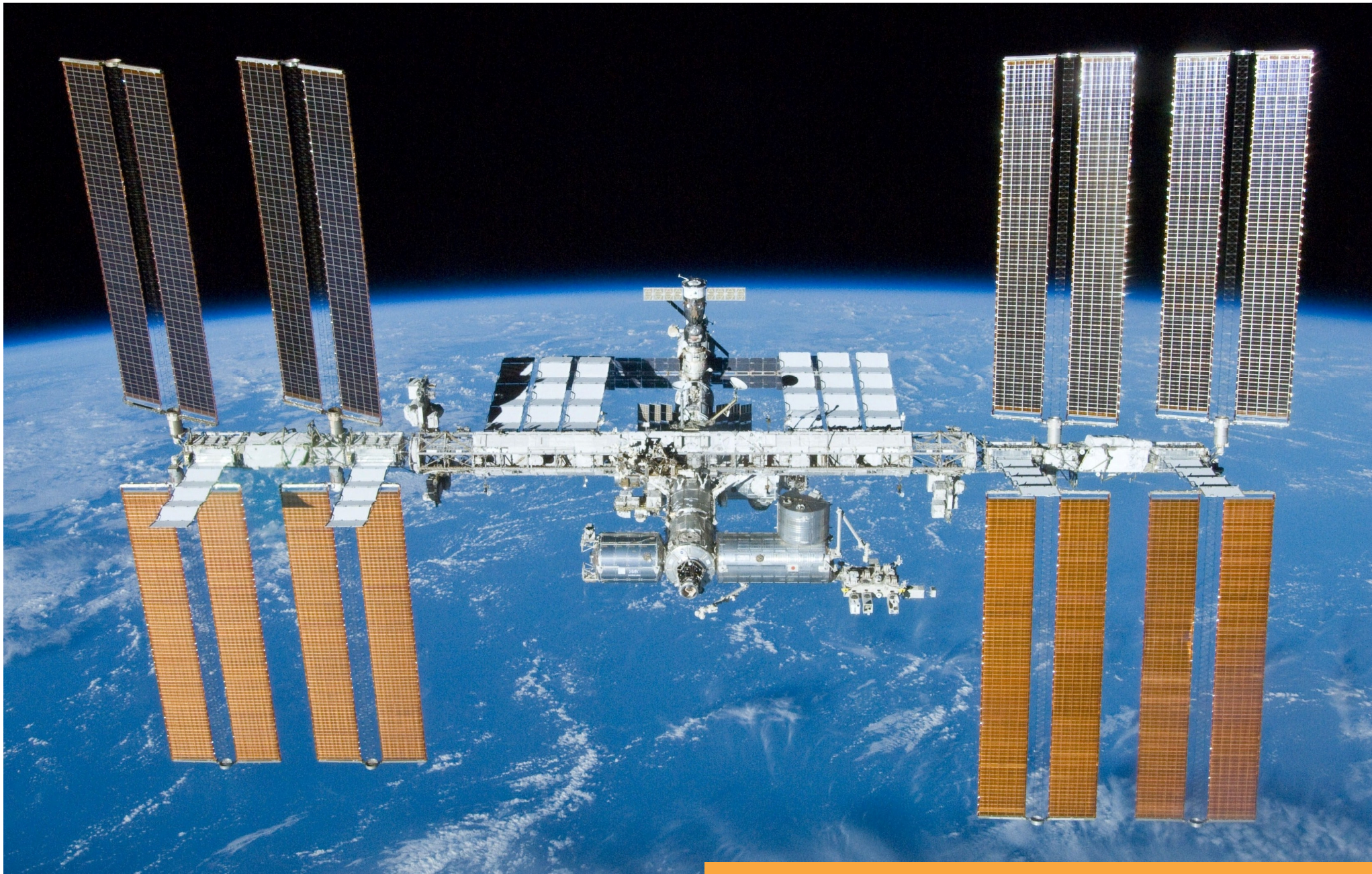
start of μg

end of μg

100 Km ~ end of atmosphere (traces)

Sounding Rockets





By NASA/Crew of STS-132 - <http://spaceflight.nasa.gov/gallery/images/shuttle/sts-132/hires/s132e012208.jpg> (<http://spaceflight.nasa.gov/gallery/images/shuttle/sts-132/html/s132e012208.html>), Public Domain, <https://commons.wikimedia.org/w/index.php?curid=10561008>

ISS
International Space Station



Call sign Alpha, Station
Crew Fully crewed 6
Currently aboard 6 (Expedition 42)
Launch 1998
Launch pad Baikonur 1/5 and 81/23
Kennedy LC-39
Mass approximately 450,000 kg
Length 72.8 m
Width 108.5 m
Height ≈ 20 m
Pressurised volume 837 m³ (21/3/2011)

Atmospheric pressure 101.3 kPa
Perigee 416 km
Apogee 425 km
Orbital inclination 51.64 degrees
Average speed 7.66 km/s (27,600 km/h)
Orbital period 92.84 minutes
Orbit epoch 15 November 2014
Days in orbit 5851 (27 November)
Days occupied 5138 (27 November)
Number of orbits 91478
Orbital decay 2 km/month

Energy & al.

Solar Array Length: 73 meters

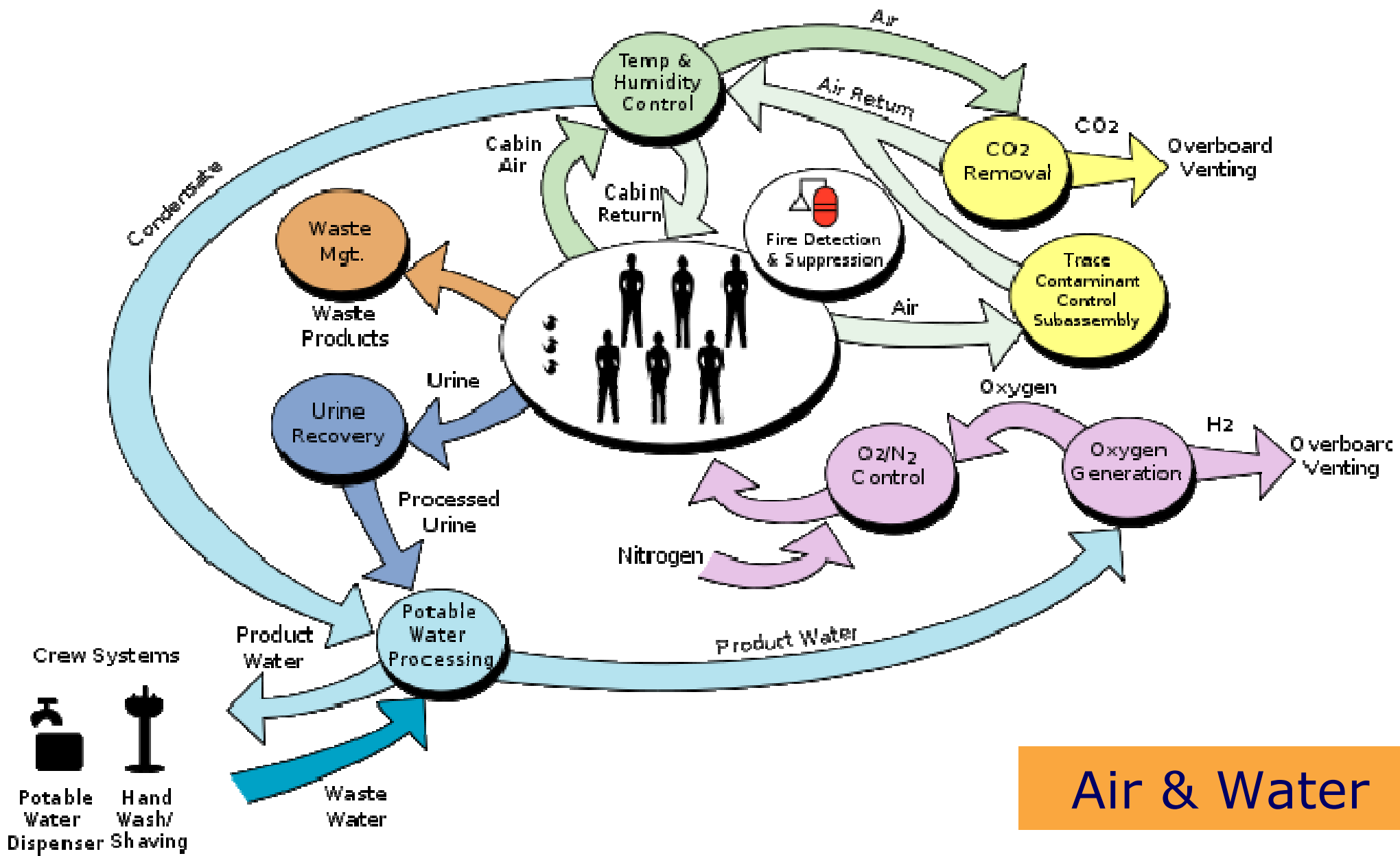
Habitable Volume: 388 cubic meters

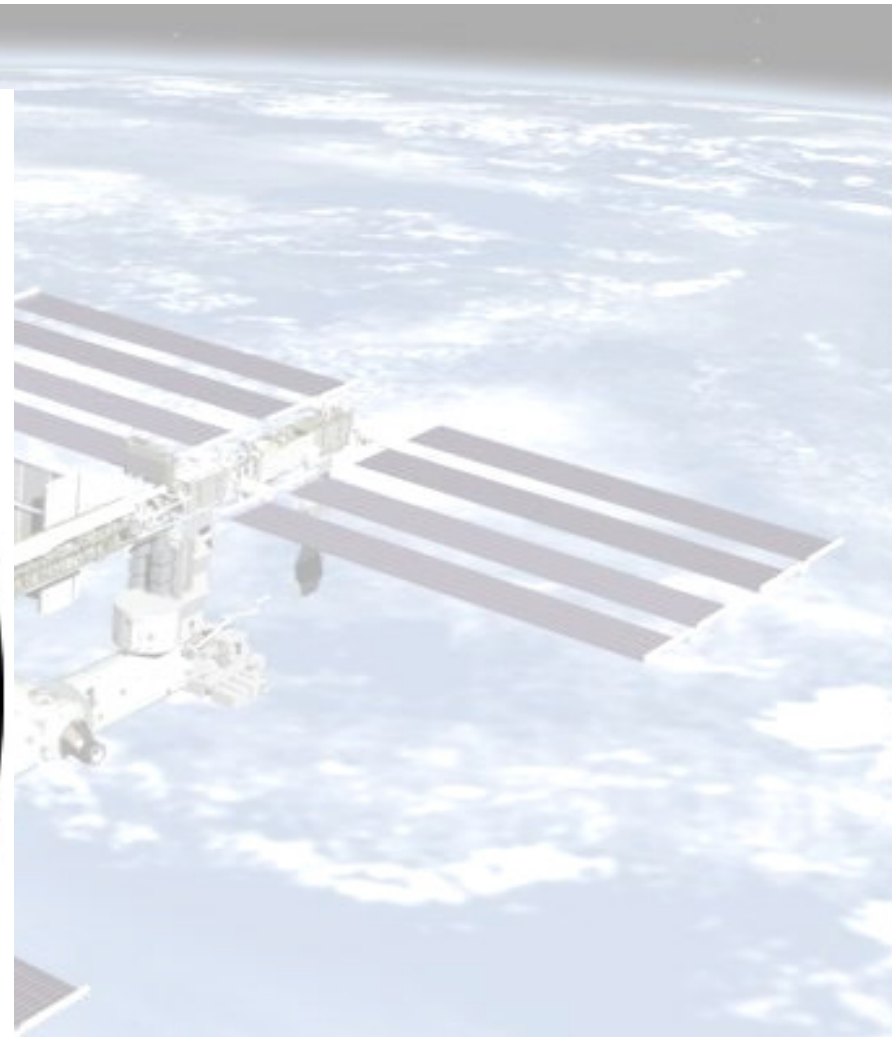
Power Generation: 8 solar arrays = 84 kilowatts

Lines of Computer Code: approximately 2.3 million


Last updated in November 3, 2014

ISS Facts and Figures





**FLEX-ICE-GA
STICKER**

The background of the slide is a photograph of the International Space Station (ISS) in orbit above the Earth. The station's complex structure, including multiple modules and large solar panel arrays, is clearly visible against the blue and white clouds of the planet. The text is overlaid on this image.

OK, now you are ready to participate in a mission on the ISS but first you have to be trained on a particular experiment will be carried on board ...

FLEX-ICE-GA

The background of the slide is a photograph of the International Space Station (ISS) in orbit above the Earth's cloud-covered surface. The station's complex structure, including its large solar panel arrays, is clearly visible against the blue and white of the planet.

FLEX : Flame Extinguishment Experiment

ICE : Italian Combustion Experiment

GA : Green Air

Partnership ICE-GA:

AGT (small enterprise)

ASI (Italian Space Agency)

DTM (small enterprise)

IM-CNR (Istituto Motori of CNR)

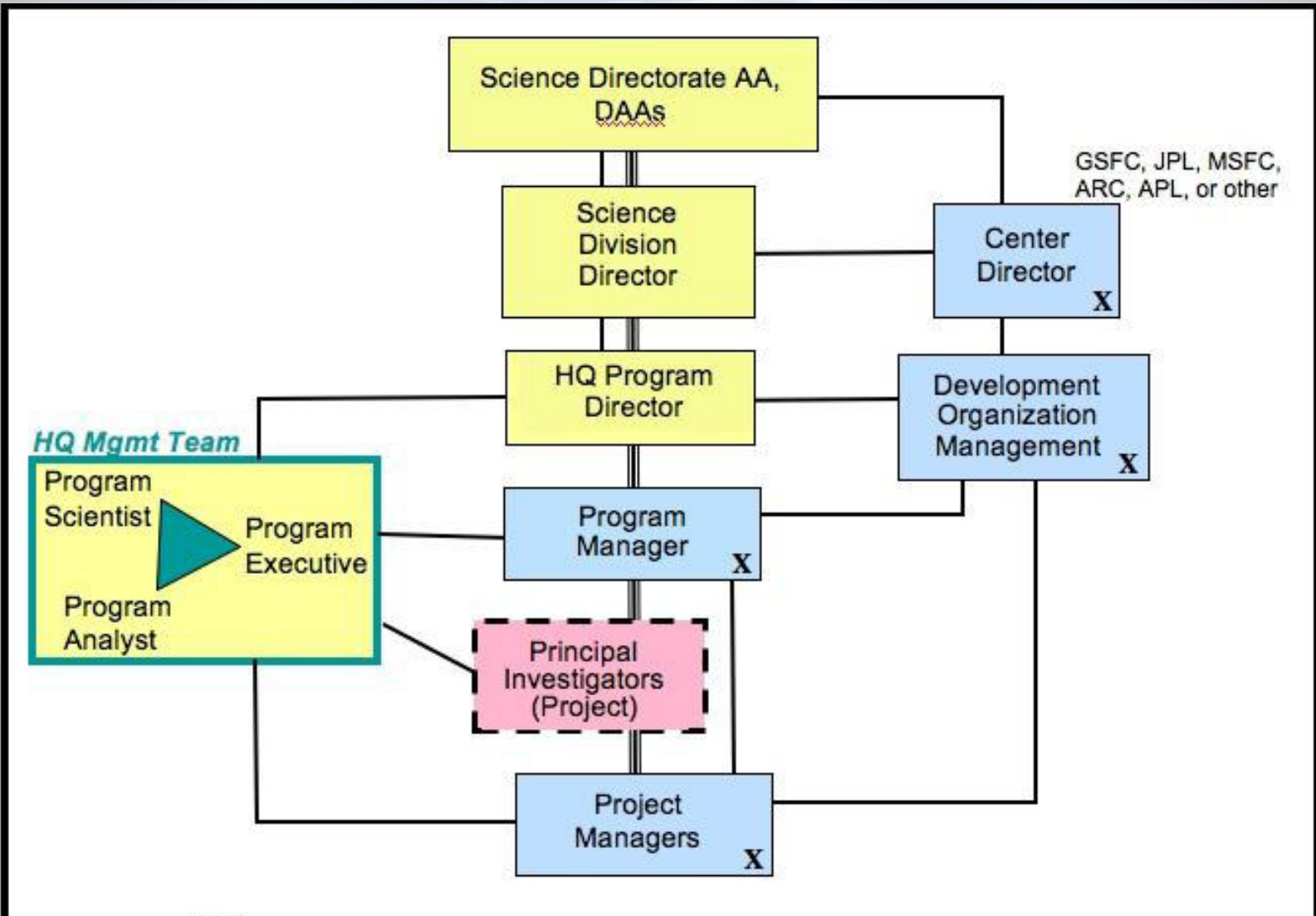
Scientific Collaboration:

ASI

IM-CNR

NASA

FLEX-ICE-GA



GSFC, JPL, MSFC, ARC, APL, or other

Key:

- ==== Programmatic Direction
- Information & Coordination
- Yellow box: Located at HQ
- Blue box with X: Located at Center
- Pink dashed box: Located at PI institution

NASA Project Structure

SCIENCE CONCEPT REVIEW OUTLINE

- i. Welcome (NASA GRC Division Chief or Program Manager)
- ii. Instructions to Science Panel (NASA Enterprise Discipline Scientist)
0. Executive Summary (PI)
 1. Goals/Objectives
 2. Proposed Space Experiment (concept diagram)
 3. Benefits (potential application)
1. Introduction and Background (PI)
 1. Description of Science
 2. Brief Historical Overview of Science
 3. Currently Active Research
 4. Current Status of Understanding
 5. Gaps in Understanding this Experiment Plans to Fill
2. PI Research Related to Proposed Space Experiment (PI)
 1. Experiments - 1g Laboratories, Drop Towers, and Aircraft
 2. Models - Numerical and Analytical
3. Proposed Space Experiment (PI)
 1. Objective and Hypothesis of Proposed Investigation
 2. Benefit to Science and Technology
 3. Flight Experiment Description
 4. Science Requirements
 5. Test Matrix
 6. Success Criteria (minimum and complete)
 7. Anticipated Results
4. Justification for Extended Duration Microgravity Environment (PI)
 1. Limitations of Terrestrial (1-g laboratory) Testing
 2. Limitations of Drop Towers and Aircraft
 3. Need for Accommodations in the ISS, Space Shuttle or Sounding Rocket
 4. Limitations of Modeling Approaches
5. Use of Data Obtained from Proposed Space Experiment (PI)
 1. Data Reduction and Analysis
 2. Model or Hypothesis Verification
6. Proposed Space Experiment Concept (PS or PI)
 1. Description of Experiment Concept (cartoon and block diagrams)
 2. Measurements and Diagnostics Required
 3. Experiment Procedure
7. Science Plan to RDR (PI)
 1. Identify Critical Tasks and Plans for Resolution
 2. Other Science Activities
8. Summary (PI)
9. Engineering Plan to RDR (PM)
 1. Identify Critical Engineering Feasibility Issues
 2. Develop Plan for Resolution of Engineering Feasibility Issues
 3. Develop Schedule and Costs
10. Rough Order of Magnitude Schedule and Costs to Flight (PM)
11. Science Panel Caucus (PS to attend as an observer and answer questions)
12. Science Panel Feedback to PI
13. Concluding Remarks (NASA Enterprise Discipline Scientist)

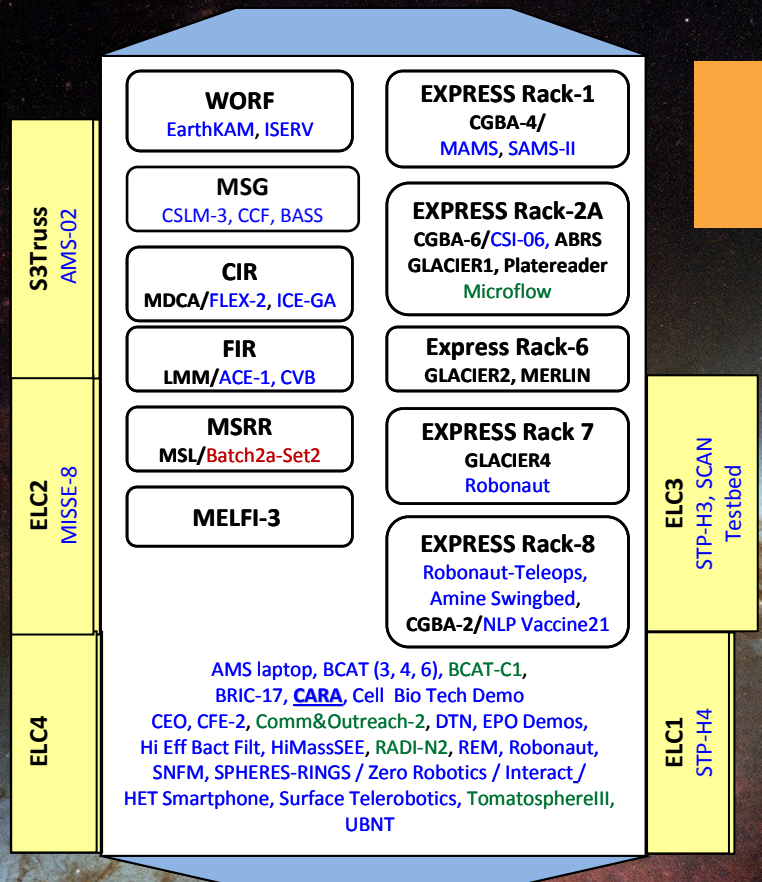
SCIENCE REQUIREMENTS DOCUMENT OUTLINE

i	SIGNATURE PAGE
ii	NOMENCLATURE
iii	ACRONYMS
iv	TABLE OF CONTENTS
v	LIST OF TABLES
vi	LIST OF FIGURES
0.0	EXECUTIVE SUMMARY
1.0	INTRODUCTION AND BACKGROUND
1.1	Brief Overview of Scientific Topic
1.2	Brief Literature Survey
1.3	Current Status of Understand
1.4	Knowledge Still Lacking
2.0	PFS RELATED RESEARCH AND PROPOGED SPACE EXPERIMENT
2.1	Experiments - 1g Laboratories, Drop Towers, and Aircraft
2.2	Models - Numerical and Analytical
2.3	Objective and Hypothesis of Proposed Investigation
2.4	Flight Experiment Description and Concept
2.5	Anticipated Knowledge to be Gained, Value, and Application
3.0	JUSTIFICATION FOR EXTENDED DURATION MICROGRAVITY ENVIRONMENT
3.1	Limitations of Terrestrial (1g laboratory) Testing
3.2	Limitations of Drop Towers and Aircraft
3.3	Need for Accommodations in the ISS, Space Shuttle or Sounding Rocket
3.4	Limitations of Modeling Approaches
4.0	EXPERIMENT PLAN
4.1	Flight Experiment Procedure
4.2	Flight Experiment Plan and Test Matrix
4.3	Postflight Data Handling and Analysis
4.4	Ground Test Plan
4.5	Mathematical Modeling
5.0	EXPERIMENT REQUIREMENTS
5.1	Science Requirements Summary Table
5.2	Test Sample
5.3	Experiment Chamber
5.4	Temperature Measurement and Control
5.4.1	Range, Accuracy and Response Rate
5.4.2	Location and Number of Sensors
5.4.3	Sampling Rate
5.5	Pressure Measurement and Control
5.6	Flow Rate
5.7	Imaging
5.7.1	Type
5.7.2	Frame Rate
5.7.3	Field of View and Resolution

Increment 35/36 Investigation Locations

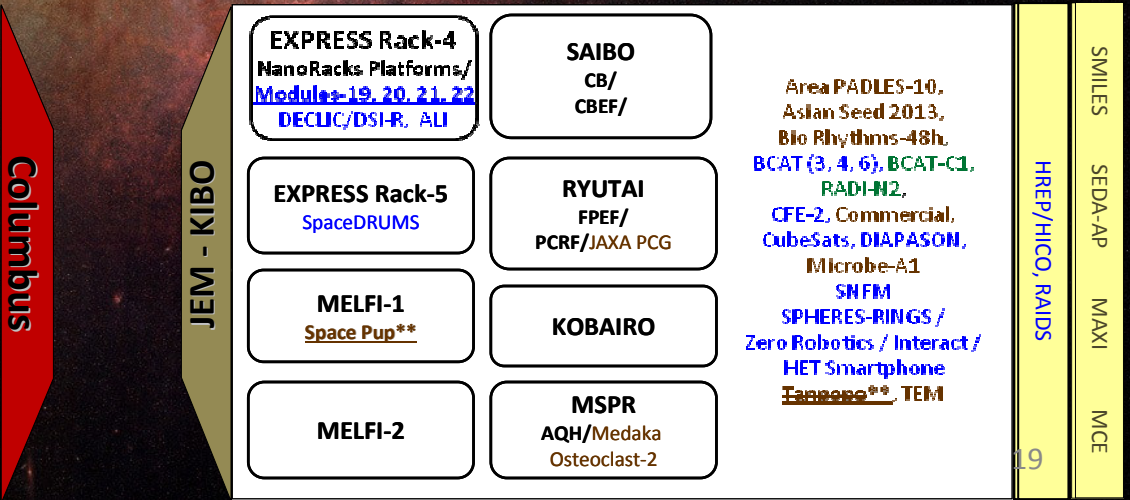
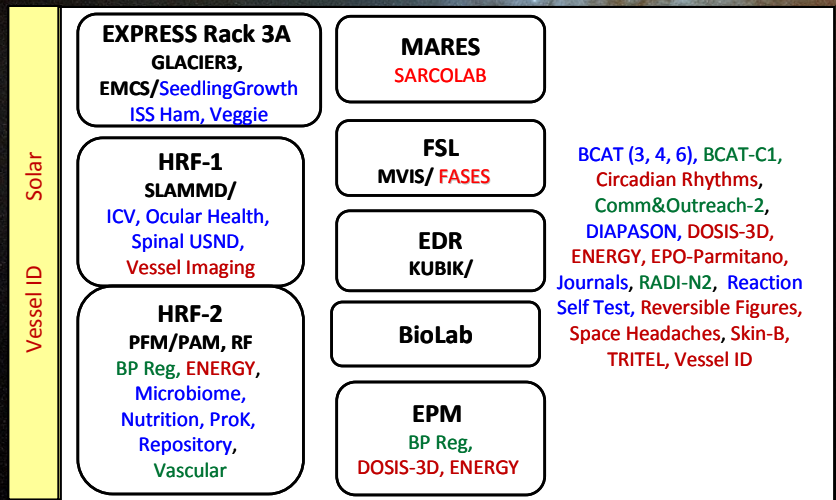
- Facility Acronyms**
- CB – Clean Bench
 - CBEF – Cell Biology Experiment Facility
 - CGBA – Commercial Generic Biology Apparatus
 - CIR - Combustion Integrated Rack
 - EDR - European Drawer Rack
 - EMCS - European Modular Cultivation System
 - EPM - European Physiology Module
 - FIR – Fluids Integrated Rack
 - FPEF – Fluid Physics Experiment Facility
 - FSL – Fluid Sciences Lab
 - HRF – Human Research Facility
 - KUBIK - Incubator
 - LMM – Light Microscopy Module
 - MARES – Muscle Atrophy Research and Ex Sys
 - MDCA – Multi-Drop Combustion Apparatus
 - MELFI – Minus Eighty deg. Laboratory Freezer
 - MSG – Microgravity Sciences Glove box
 - MSL – Material Science Lab
 - MSPR – Multi purpose Small Payload Rack
 - MSRR - Materials Science Research Rack
 - MVIS – Microgravity Vibration Isolation System
 - PCRF – Protein Crystallization Research Facility
 - WORF - Window Observation Research Facility

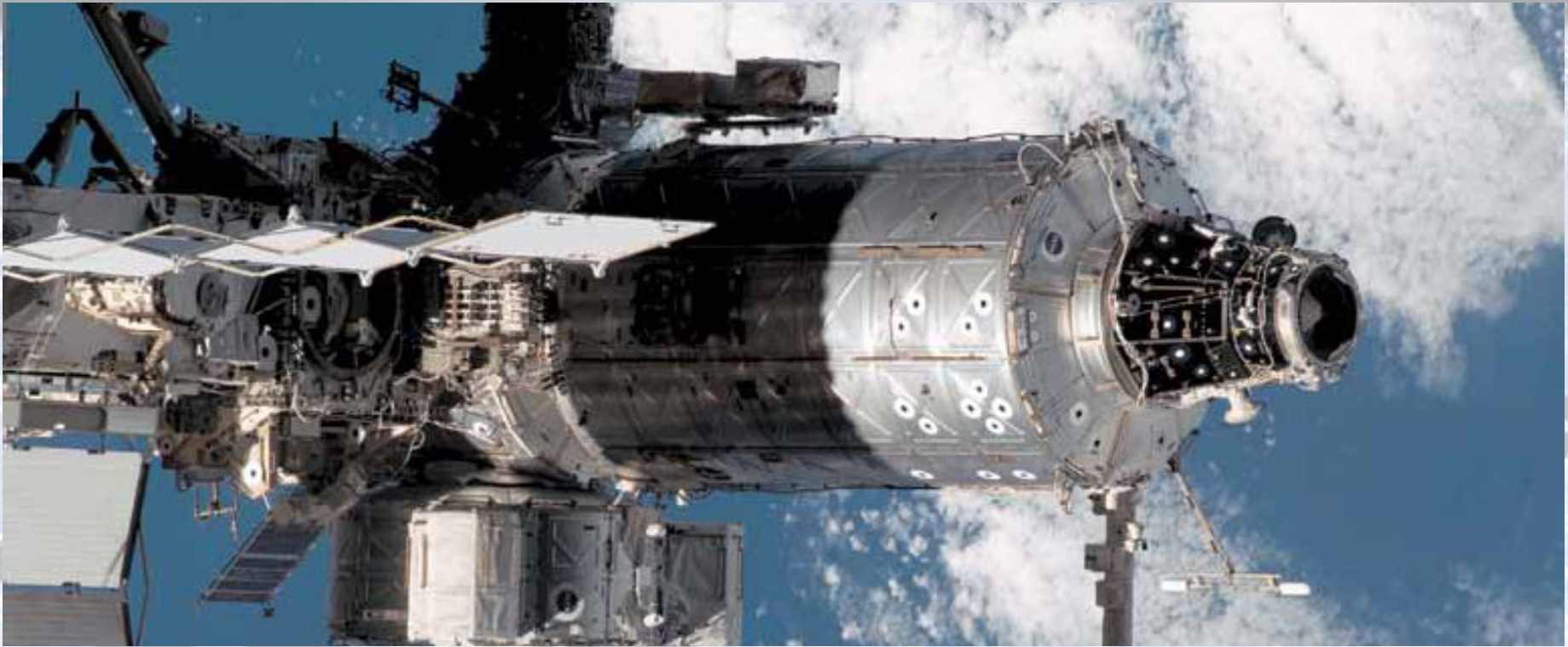
External Facility / Payload



Investigation Locations

SpX-2 Investigations included





**Fluids
Integrated Rack
(FIR)**



A complementary fluid physics research facility designed to accommodate a wide variety of micro-gravity experiments.

**Materials Science
Research Rack-1
(MSRR-1)**



Accommodates studies of many different types of materials.

**Window Observational
Research Facility
(WORF)**



Provides a facility for Earth science research using the Destiny science window on the ISS.

**Minus Eighty-Degree
Laboratory Freezer fo
ISS (MELFI-2)**



A refrigerator/freezer for biological and life science samples.

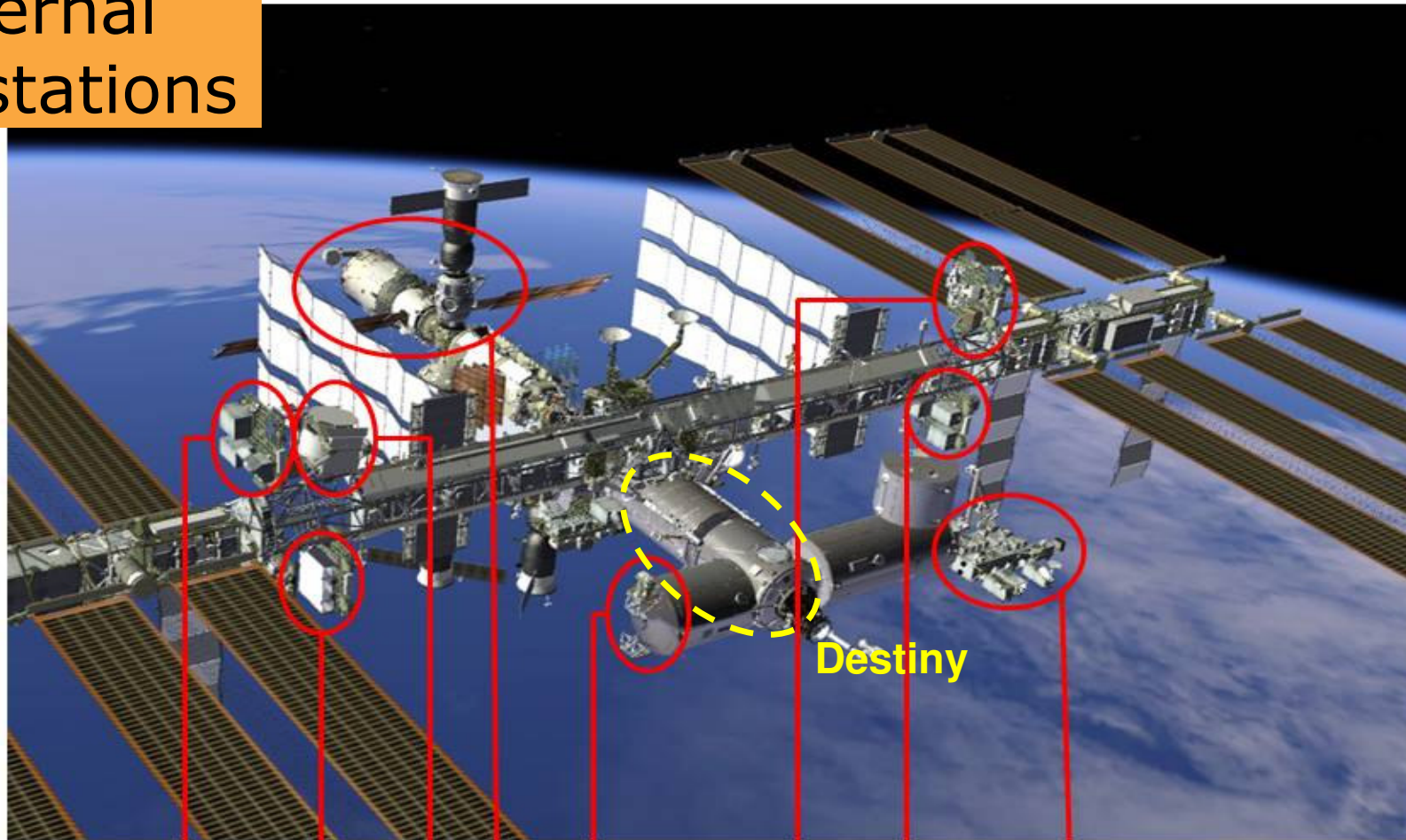
**Combustion
Integrated Rack
(CIR)**



Used to perform sustained, systematic combustion experiments in microgravity.

Destiny Racks

External Workstations



ELC-2
MISSE-8

ELC-4
RRM-P2

AMS

Columbus-EPF
Vessel ID, Solar
HDEV, NCFBX

ELC-3
STP-H3,
SCAN
Testbed

ELC-1
STP-H4
OPALS

JEM-EF
SMILES, SEDA-AP
MAXI, MCE
HREP-HICO RAIDS

External Workstations (9) on the Russian Service Module

EXPOSE-R2 on Zvezda

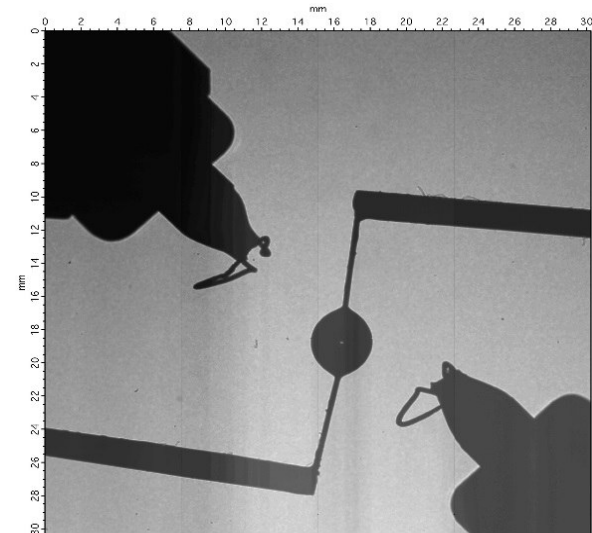
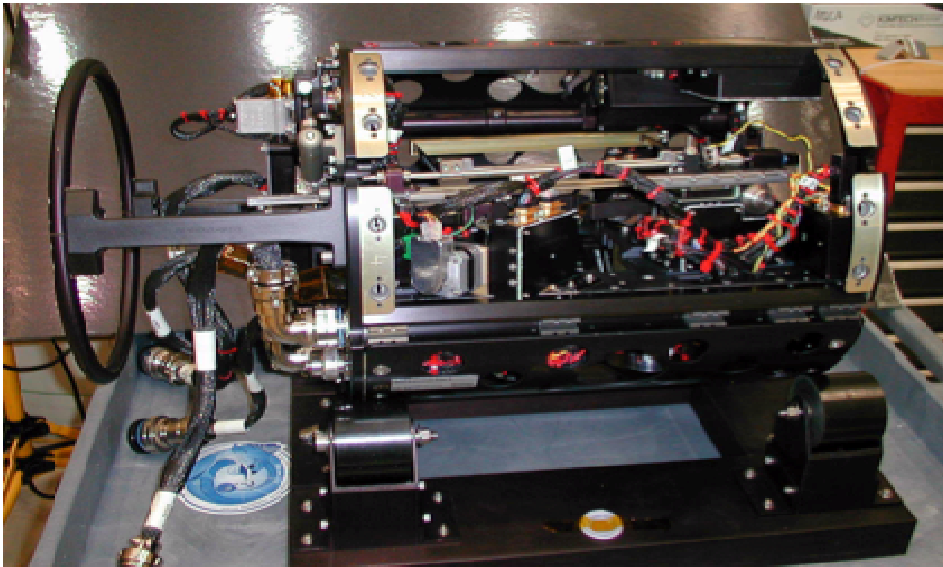
- Launched to ISS in Nov 2008 (Shuttle Endeavor)
- Operational since Mar 2009
- Multi-purpose facility for housing a range of fundamental science and spacecraft fire safety experiments
- CIR consists of:
 - 90 liter combustion chamber
 - Fuel Oxidizer Mixing Apparatus (FOMA)
 - Passive Rack Acceleration System
- Experiment-specific hardware inserted into the CIR combustion chamber
 - Multi-User Droplet Combustion Apparatus (MDCA)
- Flame Extinguishment Experiment (FLEX) is the first experiment
 - Droplet combustion experiment
 - FLEX-2
 - FLEX-2J: NASA/JAXA/Nihon Univ./Yamaguchi Univ. collaboration
 - FLEX-ICE-GA: NASA/ASI/Istituto Motori-CNR collaboration



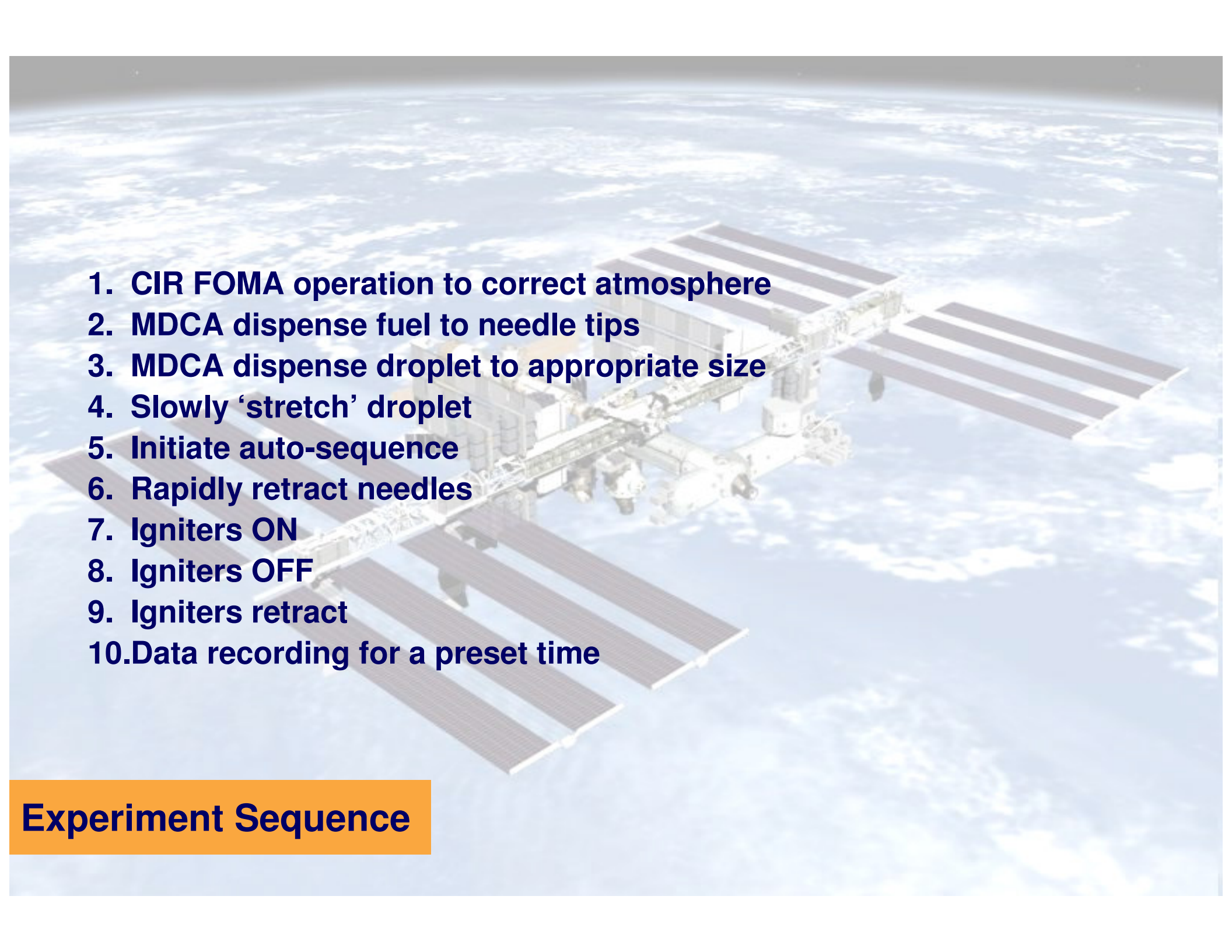
ISS018E018365

The Combustion Integrated Rack (CIR)

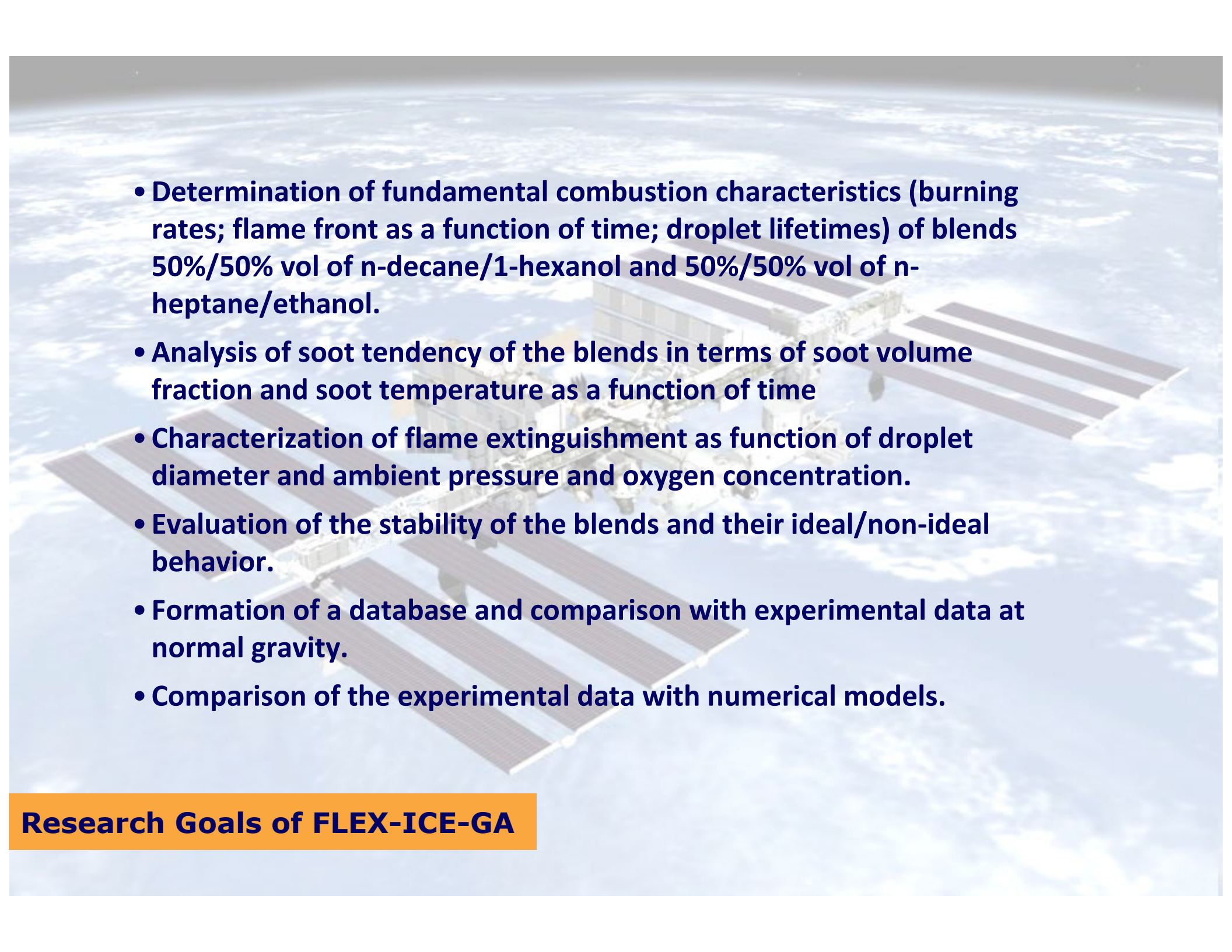
- Provides capability to store, dispense and deploy free-floating and fiber-supported droplets in microgravity.
- Fuel stored in two (crew-replaceable) syringe connected to the two fuel needles
- Ignition from two horizontally opposed hot-wire igniters



Multi-User Droplet Combustion Apparatus (MDCA)

- 
- The background of the slide is a photograph of the International Space Station (ISS) in orbit above the Earth's cloud-covered surface. The station's complex structure, including its truss, modules, and large solar panel arrays, is clearly visible against the bright blue and white of the planet.
1. CIR FOMA operation to correct atmosphere
 2. MDCA dispense fuel to needle tips
 3. MDCA dispense droplet to appropriate size
 4. Slowly 'stretch' droplet
 5. Initiate auto-sequence
 6. Rapidly retract needles
 7. Igniters ON
 8. Igniters OFF
 9. Igniters retract
 10. Data recording for a preset time

Experiment Sequence

- 
- The background of the slide is a photograph of the International Space Station (ISS) in orbit above Earth's cloud-covered surface. The station's complex structure, including multiple modules and large solar panel arrays, is clearly visible against the bright blue and white of the planet.
- **Determination of fundamental combustion characteristics (burning rates; flame front as a function of time; droplet lifetimes) of blends 50%/50% vol of n-decane/1-hexanol and 50%/50% vol of n-heptane/ethanol.**
 - **Analysis of soot tendency of the blends in terms of soot volume fraction and soot temperature as a function of time**
 - **Characterization of flame extinguishment as function of droplet diameter and ambient pressure and oxygen concentration.**
 - **Evaluation of the stability of the blends and their ideal/non-ideal behavior.**
 - **Formation of a database and comparison with experimental data at normal gravity.**
 - **Comparison of the experimental data with numerical models.**

Research Goals of FLEX-ICE-GA

Experiment Diagnostics

Backlit view of droplet:

- Droplet size, soot shell dynamics, soot volume fraction

Flame luminosity filtered for OH^* (308 nm)

- Flame size, relative intensity, flame dynamics

Color camera view of flame

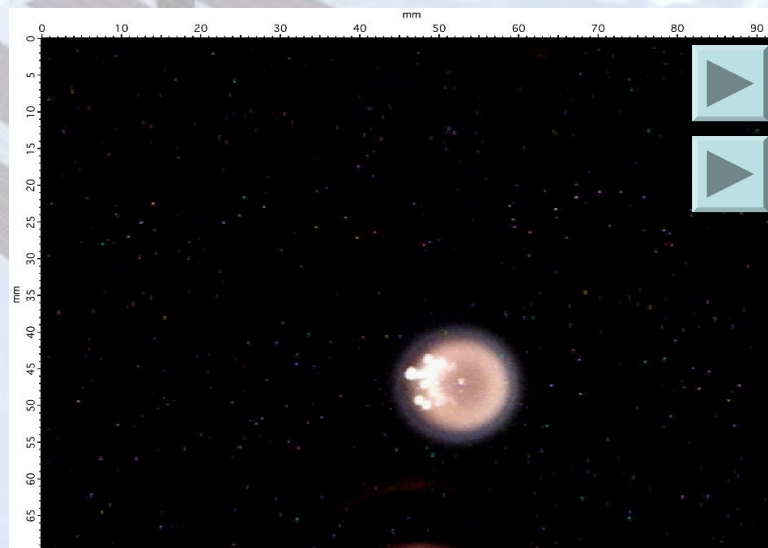
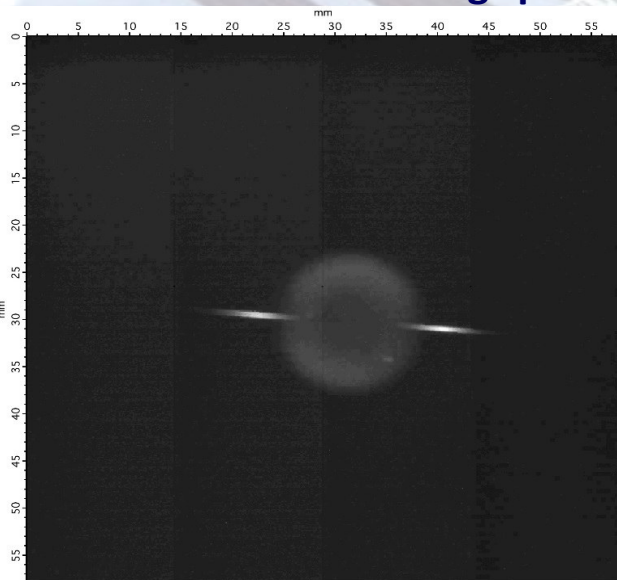
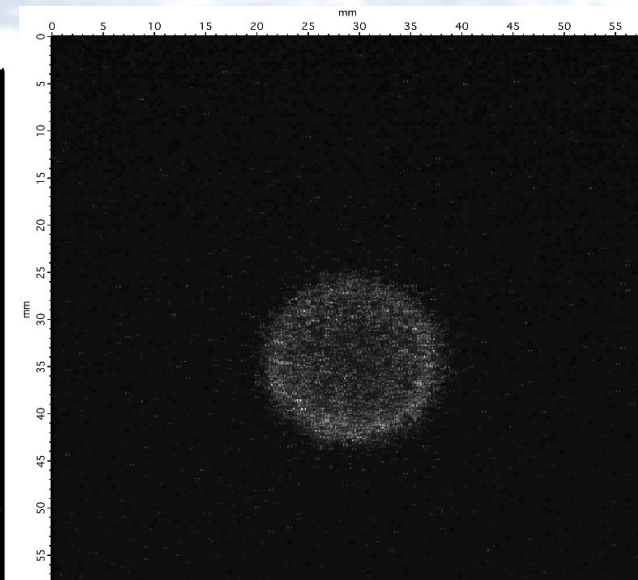
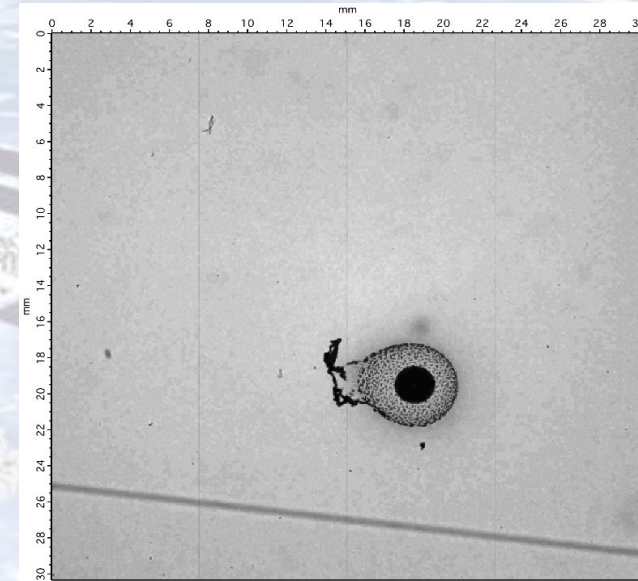
- What the eye would see
- Flame size, flame dynamics, flame color (yellow, blue)

IR-filtered view of flame/fiber

- Estimate of flame temperature (for fiber-supported tests)
- Estimate of soot temperature (in combination with soot volume fraction)

Radiometric views of flame

- Wide band radiometer
- Narrow band looking specifically at H_2O emission





Certification Completion

THE
PATRIOT

has successfully

completed the course
L: BASICS OF C

on 08/02/10

Lauren Leo

Director, Workforce Management and Development Division,
Office of Human Capital Management

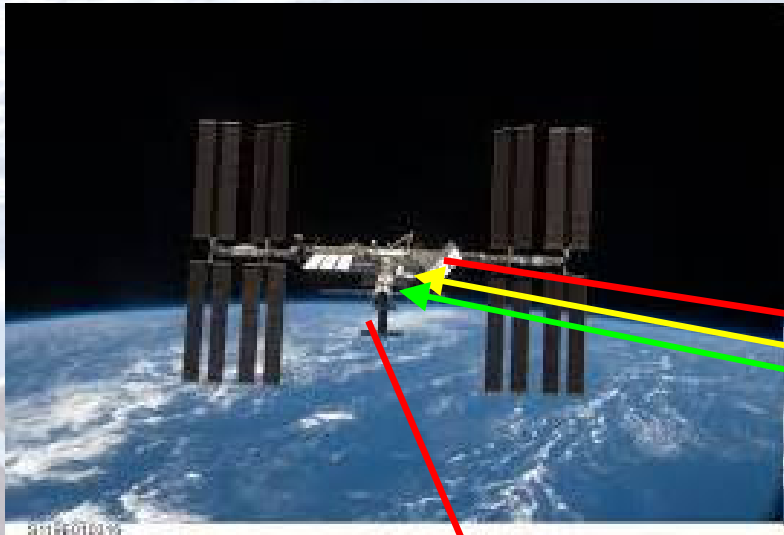
National Aeronautics and Space Administration

12

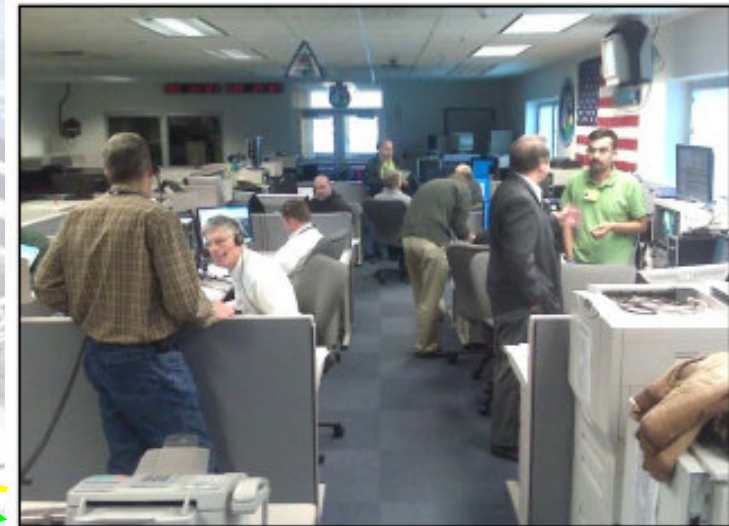
NASA as

All experiment operations controlled at *John H. Glenn Research Center (GRC) Telescience Support Center (TSC) Cleveland, Ohio.*

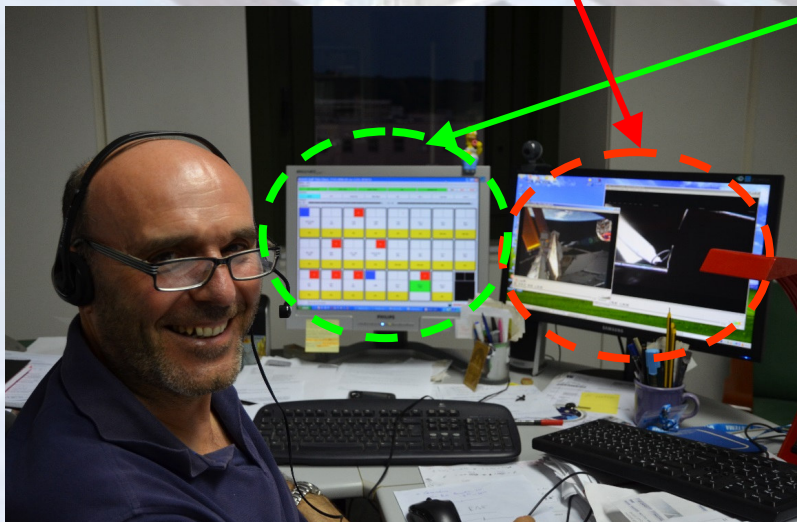
- Video/audio link to external PI



Experiment Operation



*Fluids & Combustion Facility
Mission Operations at TSC,
Building 333*



Istituto Motori - CNR, Napoli 30/7/2013 00:12

23 Voice Channel

1 dedicated to FLEX-ICE-GA

4 Webcam

2 (at least) internal

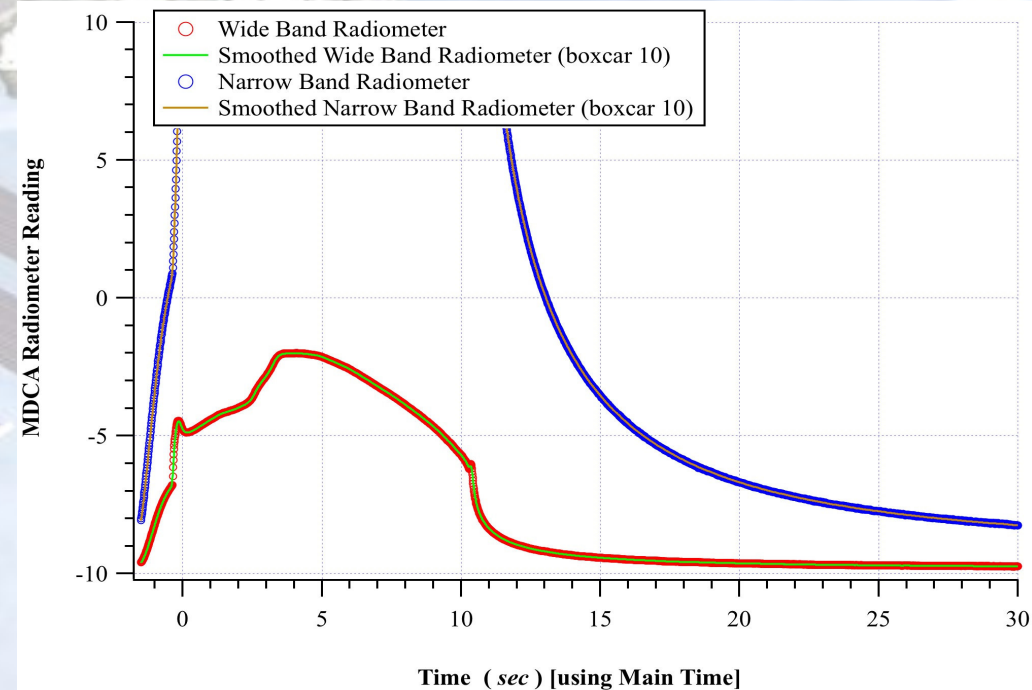
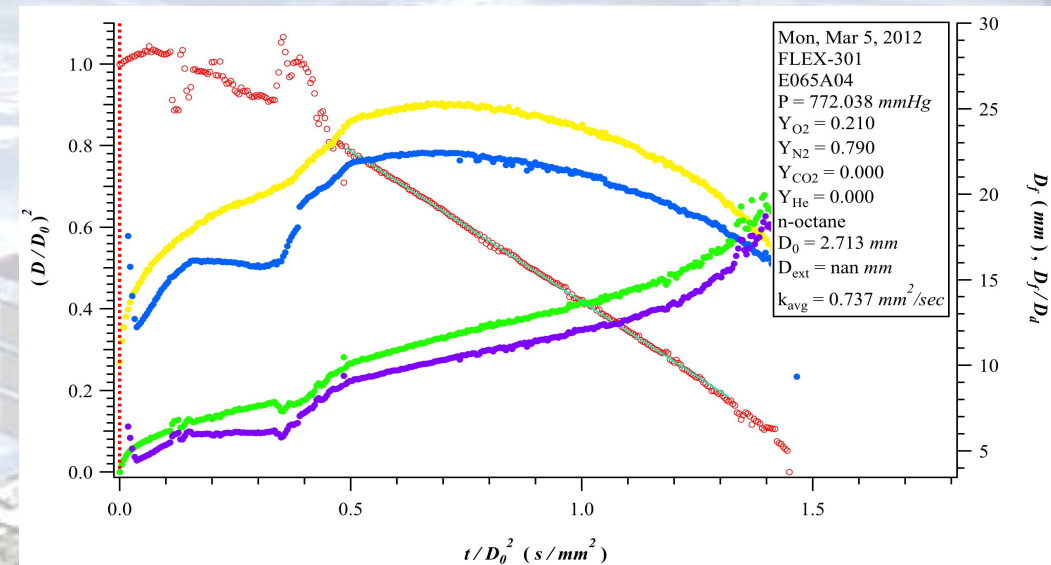
1-2 external

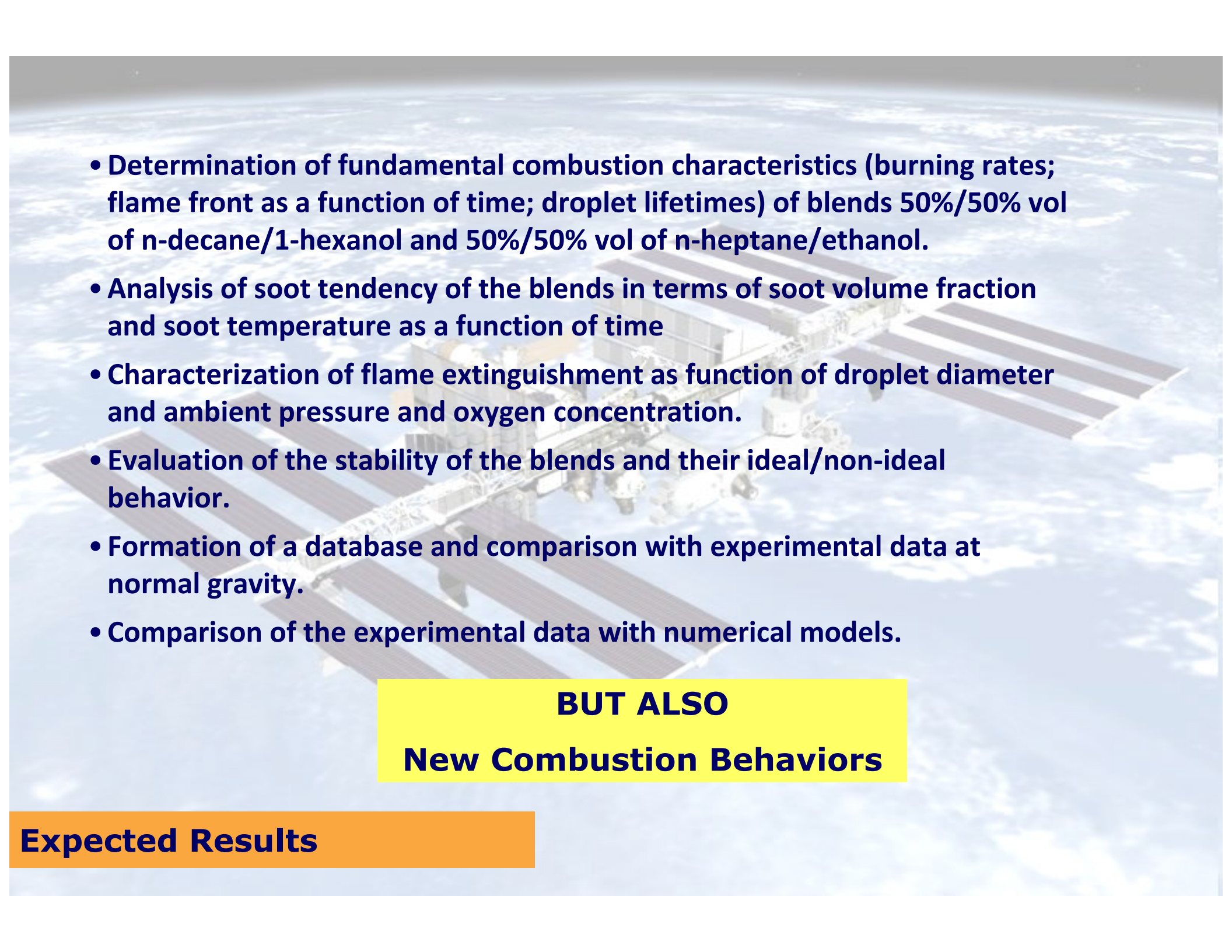


Istituto Motori - CNR, Napoli 30/7/2013 00:12

Experiment Methodology

- 4 – 10 Test points (droplet combustion tests per day)
- 1 or 2 ambient atmospheric conditions per test day
- Approximately 1-2 test day per week
- Real time experiment control at NASA GRC TSC
- Real-time audio/video feed to ASI/Istituto-Motori
- Approximately 1 day required for data downlink
- Approximately 1 week for data decompression/formatting
- Collaborative data analysis between Istituto-Motori and NASA
- 1,5 months of experiment operations
- Schedule allows for modification of test matrix between test point days
- Examine parameter space of interest based on previous results

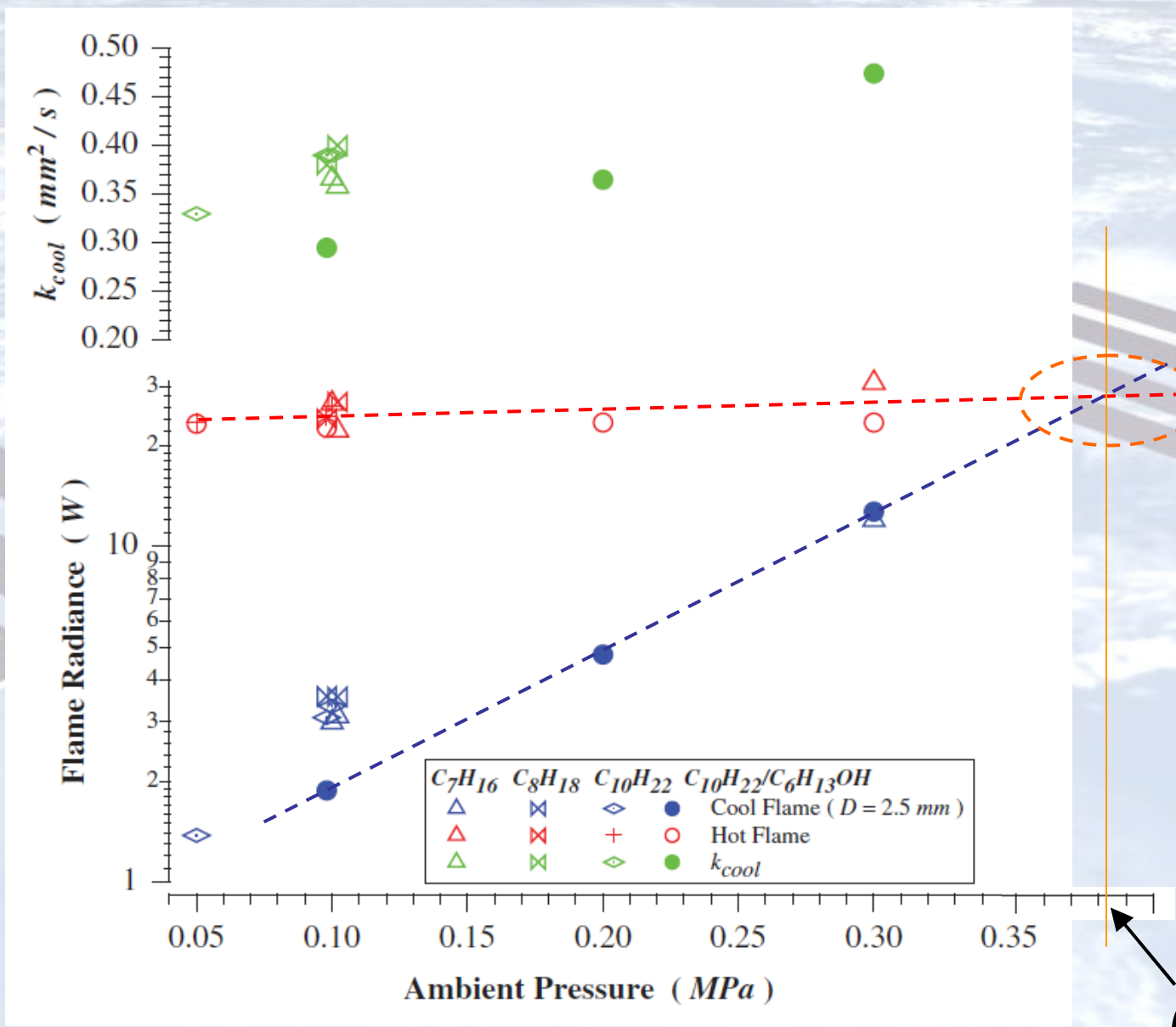


- 
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 - **Comparison of the experimental data with numerical models.**

BUT ALSO

New Combustion Behaviors

Expected Results



0.382 MPa

D. L. Dietrich, R. Calabria, P. Massoli, V. Nayagam & F. A. Williams (2017) Experimental Observations of the Low-Temperature Burning of Decane/Hexanol Droplets in Microgravity, Combustion Science and Technology, 189:3, 520-554, DOI: 10.1080/00102202.2016.1225730

The image shows the International Space Station (ISS) in orbit above the Earth. The station's complex structure, including multiple modules and large solar panel arrays, is clearly visible against the blue and white clouds of the planet. The text is overlaid on the central part of the image.

Thank you for your attention



..... first part

now.... the future ...

Tre italiani nello spazio con Virgin Galactic nel 2020

Faranno esperimenti su corpo umano e carburanti verdi



Redazione ANSA 04 ottobre 2019 08:58  Scrivi alla redazione  Stampa



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A wide-angle photograph of the Virgin Galactic launch site at dusk. The scene is dominated by a long, straight runway that recedes into the distance, lined with blue lights. In the background, a range of rugged, snow-capped mountains rises against a dark, twilight sky. Several wind turbines are visible on the left side of the image. The overall atmosphere is serene and industrial.

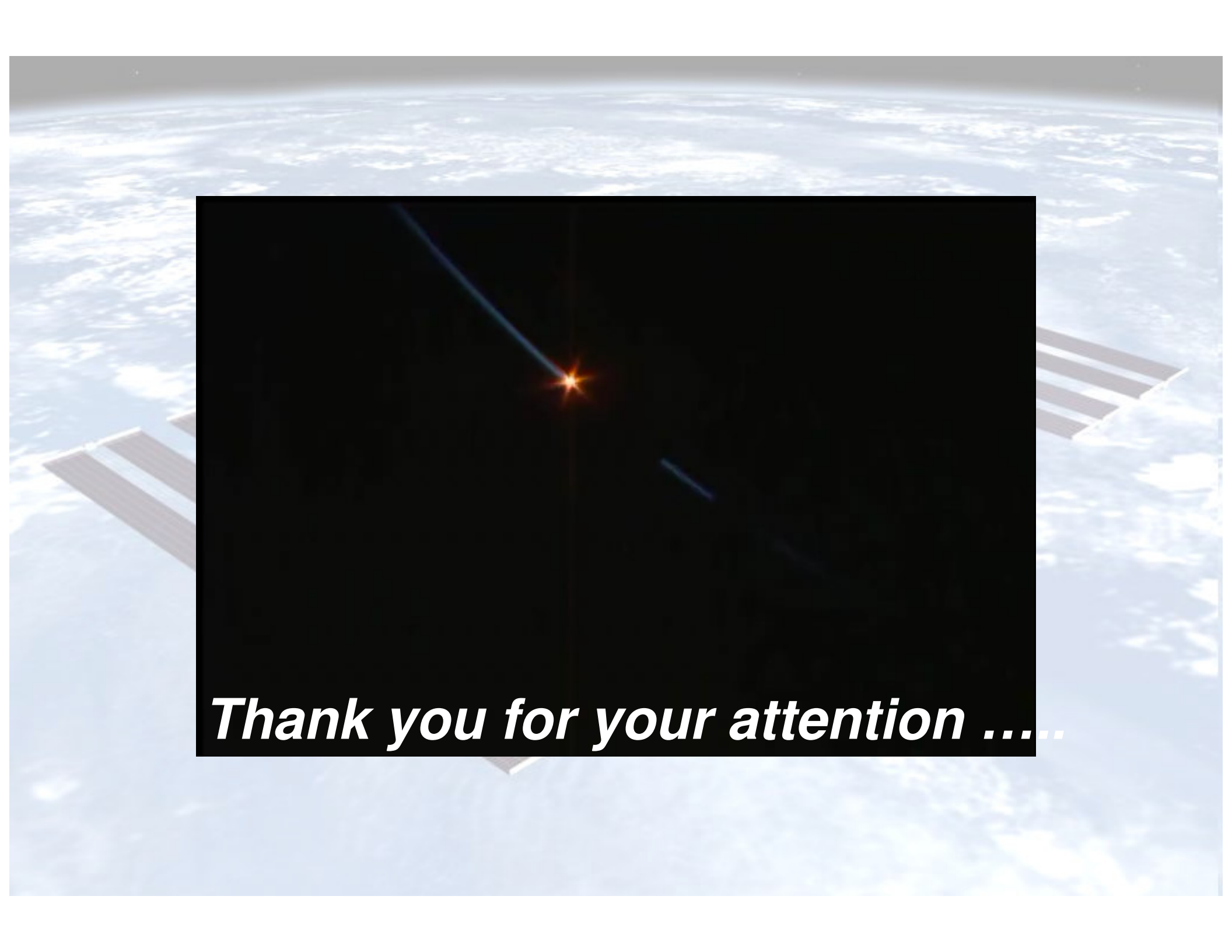
VIRGIN SPACESHIP UNITY SECOND SPACEFLIGHT

Comunicato Ansa 3 ottobre 2019

Nel 2020 tre ricercatori italiani avranno la possibilità di condurre esperimenti nello spazio durante un volo suborbitale a bordo dello spaziosplano SpaceShipTwo di Virgin Galactic: lo stabilisce l'accordo siglato nella sede dell'ambasciata italiana a Washington dalla compagnia del magnate Richard Branson con l'Aeronautica Militare Italiana. E' la prima volta che un dipartimento governativo finanzia un volo umano a scopo di ricerca scientifica su un veicolo spaziale commerciale.

La missione porterà nello spazio tre specialisti di carico italiani e uno 'scaffale' di esperimenti scientifici. L'Aeronautica militare e il team della Virgin Galactic stanno già lavorando con il Consiglio nazionale delle ricerche (Cnr) per progettare i carichi degli esperimenti.

"Durante il volo, dopo lo spegnimento del motore del razzo, i ricercatori si sganceranno dai sedili ed eseguiranno le azioni necessarie per completare ciascun esperimento nel giro di alcuni minuti a gravità zero", si legge in una nota congiunta. "La compagnia si occuperà dell'addestramento e della preparazione dei ricercatori, in modo che siano pienamente equipaggiati per compiere il loro lavoro come specialisti della missione sul volo, e fornirà in loco il supporto pre-volo per ciascuno dei carichi sperimentali". Tra questi saranno inclusi strumenti per misurare gli effetti biologici sul corpo umano della transizione dalla gravità alla microgravità. Altre attrezzature potrebbero essere usate per studiare meglio la chimica dei carburanti verdi.

The image is a composite. The background is a view of Earth from space, showing a blue and white horizon with clouds. In the center, there is a black rectangular area. Inside this black area, there is a bright orange star with a lens flare effect, and several thin blue lines radiating from it. At the bottom of the black area, there is white text that reads "Thank you for your attention".

Thank you for your attention